

Bundesamt für Zivilluftfahrt (BAZL) Office fédéral de l'aviation civile (OFAC) Ufficio federale deil'aviazione civile (UFAC) Federal Office for Civil Aviation (FOCA)

3003 Bern/Berne/Berna

Lufttüchtigkeitsanweisung Consigne de navigabilité Prescrizione di aeronavigabilità Airworthiness directive

HB 91-146

Inkraftsetzung Date en vigueur Entrata in vigore 14.5.1991

Effective Date

Betroffene Muster – Types concernés – Applicabilità – Models affected

FFA Flugzeugwerke Altenrhein AG, AS202/15, AS202/15-1, AS202/18, AS202/18-1, serial number (S/N) 001--135, excluding S/N 105 and 106 and airplanes on which the black elevator drive shaft assy has already been replaced since 1988 by the grey painted one

Anlass/Massnahmen - Objet/Mesures - Oggetto/Provvedimenti - Subject/Action

Elevator drive shaft assembly - Corrosion in the tube including the welding inside edge - Removal of the elevator drive shaft assembly and installation of a shaft assy with improved surface treatment

Fristen – Délais – Scadenza – Compliance

Within 10 hours time in service, but not later than June 15, 1991

Herkunft – Provenance – Provenienza – Origin

FFA Flugzeugwerke

Bezugnahme - Référence - Riferimento - Reference

FFA Service Bulletin AS 202 Bravo No. 41A (attached)

Bemerkungen - Observations - Osservazioni - Remarks

Supersedes HB 91-001

Rechtsmittelbelehrung	siehe	ТΜ	Nr.	
Voies de droit	voir	CT	no.	F 80.015-10
Rimedi giuridici	vedi	СТ	no.	1- 00.013-10
Right to appeal	see	TC	no.	

Seite		von	
Page	4	de	5
Pagina	I	di -	
Page		of	

HB 91-146)		
	A	S202 BRAVO
		May 1991 No 41A
	SERVICE BULLET	IN
	Mandatory	Page 1 of 4 Supersedes SB No 41
	ELEVATOR DRIVE SHAFT ASSY	
1. Planning Int	Formation	

1.1 Effectivity - Models: AS 202/15, AS 202/15-1 AS 202/18, AS 202/18-1

> - Ser. Nos. S/N 001 to 135 (Excluding S/N 105 and 106 and Airplanes

- 1.2 Reason:
- 1.3 Description:
- 1.4 Compliance:
- 1.5 Approval:
- 1.6 Manpower:

1.8

- 1.7 Material: Available from FFA
 - Tooling: No special tools are required
- 1.9 Weight and Balance: not affected
- 1.10 Electrical load data: not affected
- 1.11 Reference to other publications:
- AMM / IPC, chapter 27-31-04, 27-31-05 55-20-01, 50-50-01 55-20-02

on which the black elevator drive shaft assy has already been replaced since

assembly and installation of shaft assy

Within 10 hours time in service but not

Corrosion in the tube including the

1988 by the grey painted one)

Removal of elevator drive shaft

with improved surface treatment

Approved by the Swiss Federal Air

Office as Airworthiness Directive

later than June 15, 1991

Approx. 5 man-hours

welding inside edge

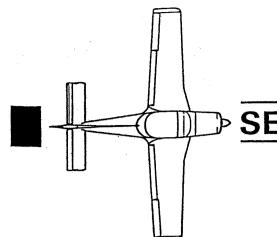
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- 2 -

(HB 91-146)



AS202 BRAVO

No 41A

SERVICE BULLETIN

Page 2 of 4

2. Accomplishment Instructions

2.1 Remove

- Remove left cover 202-17.505.11 and right cover 202-17.505.12 on the rear of the fuselage
- Disconnect the elevator steering rod from lever on the drive shaft assy
- Remove covering 202-17.479 of the bearings on the left elevator
- Disconnect the attachment of left elevator on the drive shaft assy
- Remove hinge bolts on the left elevator and carefully remove the left elevator
- Disconnect the middle hinge bolt on the drive shaft assy
- Disconnect the attachment of right elevator on the drive shaft assy and remove the drive shaft assy

2.2 Installation

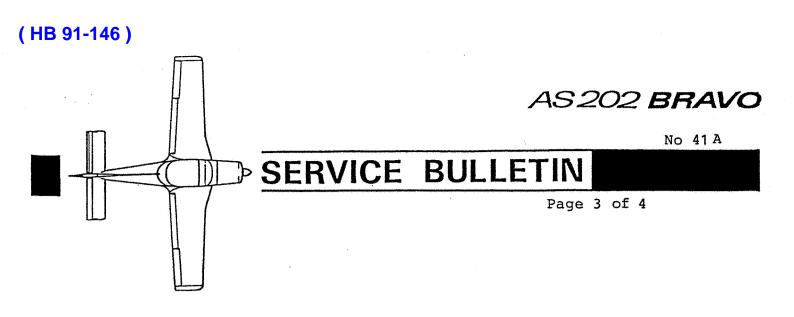
- Install the replacement drive shaft assy on the right elevator, see IPC 55-20-01 Page 0, lock the bolts with locking plate and locking wire
- Install the middle hinge bolt on the drive shaft assy, see IPC 55-50-01 Page 0, lock the castellated nut with cotter pin MS 24665-151
- Install the left elevator in position, connect the hinge bolts, see IPC 55-50-01 Page 0, lock the castellated nuts with cotter pin MS 24665-151

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- Attach the left elevator on the drive shaft assy, see IPC 55-20-01 Page 0, lock the bolts with locking plate and locking wire
- Attach the elevator steering rod on the lever of drive shaft assy, see IPC 27-31-04 Page 0 or 27-31-05 Page 0. Lock the castellated nut with cotter pin MS 24665-153
- Install the cover 202-17.479 at the bearings on the left elevator, see IPC 55-20-02 Page 0
- Install the left cover 202-17.505.11 and right cover 202-17.505.12 rear on the fuselage

2.3 <u>Test</u>

- Check the elevator deflection:

For 150 HP airplanes:

up 30° ±1° down 20° ±1°

For 180 HP airplanes: up $30^{\circ} \pm 1^{\circ}$ down $30^{\circ} \pm 1^{\circ}$

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4 -

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(HB 91-146)



3. Material

For completion of this Service Bulletin the following material is necessary:

- 1 piece Elevator drive shaft assy 202-17.450
- 1 piece Cotter pin MS 24665-153
- 2 pieces Cotter pin MS 24665-151
- 2 pieces Locking wire MS 20995-NC32 (lenght approx. 12" = 300 mm)
- 2 pieces Locking plate 202-17.475

The accomplishment of this Service Bulletin should be recorded in the technical document of the airplane.

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5-